

AD 718398

THE **BOEING** COMPANY

REV 1TR

CODE IDENT. NO. 81205

NUMBER D2-125304-1

TITLE: HUGONIOT EQUATION OF STATE OF MYLAR

ORIGINAL RELEASE DATE _____ FOR THE RELEASE DATE OF SUBSEQUENT REVISIONS, SEE THE REVISIONS SHEET. FOR LIMITATIONS IMPOSED ON THE DISTRIBUTION AND USE OF INFORMATION CONTAINED IN THIS DOCUMENT, SEE THE LIMITATIONS SHEET.

MODEL _____ PREPARED UNDER:

ISSUE NO. _____ ☐ CONTRACT NO.

ISSUE TO _____ ☐ IR&D
☐ OTHER

PREPARED BY R. D. Davis

SUPERVISED BY R. D. Davis

APPROVED BY M. F. East

APPROVED BY M. F. East

Reproduced by
**NATIONAL TECHNICAL
INFORMATION SERVICE**
Springfield, Va. 22151

DISTRIBUTION STATEMENT A

Approved for public release;
Distribution Unlimited

DDC
RECEIVED
FEB 24 1971
RECEIVED

SHEET

BIS 13 12

MYLAR

△ Mylar has been used for a variety of applications in shock loading experiments. In particular it has been utilized as thin flyer plates in damage threshold and Hugoniot measurements performed using exploding foil facilities and thus its Hugoniot "equation of state" is basic to the interpretation of these experiments. ~~Recently~~ an experimental determination of the Hugoniot of Mylar in the stress range 3 to 30 kilobars has been performed, which is, to our knowledge, the first accurate data that has been obtained for this material.

△ In the experimental method employed, a shock was initiated by impacting a flat target with a flat-nosed projectile accelerated by a compressed gas gun. The experimental arrangement is shown in Figure 1. The technique makes use of stress gauges whose principle of operation is based on the piezo-electric properties of quartz and which have been discussed in detail by R. A. Graham.¹ The target, a quartz gauge, is mounted at the end of an evacuated gun barrel and impacted by a precisely aligned flat faced specimen of Mylar attached to a projectile. The velocity of the projectile is accurately measured. The gauge signals are transmitted by co-axial cable to oscilloscopes for recording. The pin mounted ahead of the gauge is used to trigger these oscilloscopes.

In order to obtain a sample that could be mounted on a projectile, the Mylar specimen was made up of a number of layers of 0.014" thick Mylar. The layers were bonded together using a thermo plastic adhesive, Schjel-Bond.

1. R. A. Graham, F. W. Neilson, W. B. Benedick
Piezoelectric Current From Shock Loaded Quartz - A Submicrosecond
Stress Gauge, J. App. Physics

Overall sample thickness was approximately 0.25". The gauge records indicate that a jump in stress is impressed upon a specimen at the impact surface. The magnitude of this jump is calculated directly from the gauge signal as described in Reference 1.

For a one-dimensional experiment as performed here, conservation of momentum requires that

$$U_q = \frac{\sigma}{P_q D}$$

where U_q is the particle velocity imparted to the gauge taken with respect to its unstressed portion, σ is the stress, D is the shock velocity, and P_q is the initial density of the quartz. Further, if continuity of mass across the impact surface is assumed then

$$U_m = U_o - U_q$$

where U_m is the particle velocity imparted to the Mylar and U_o is the impact velocity of the projectile.

Thus, since U_o is measured, it is possible to calculate the stress and particle velocity in the Mylar if the Hugoniot for quartz is known.

The following relationships between the shock and particle velocity for quartz derived by R. A. Graham¹ were used in the analysis of the data.

For

$$0 < \sigma < 25 \text{ kbars}$$

$$D = 5.740 + 0.14 U_q \text{ (mm/}\mu\text{Sec)}$$

and for

$$25 < \sigma < 50 \text{ kbars}$$

$$D = 5.57 + 1.08 U_q \text{ (mm/}\mu\text{Sec)}$$

Figure 2 shows a typical gauge record. The effects of the layering of the Mylar sample can clearly be seen on the top of the pulse. However, it should be noted that the initial stress jump is in no way modified by the layering and the derived

Hugoniot refers only to the Mylar. The sheets of Mylar used in the laminations were 0.014" thick thus providing the rise-time of the stress jump is less than 0.2 μ Sec the impact interface can have no knowledge of the first bond prior to the gauge signal attaining its true initial amplitude.

The data obtained is tabulated in Table 1 and plotted in Figures 3, 4 and 5. The shock velocity of Mylar at zero stress, i.e., sound velocity, was measured using a pulse technique and was found to be 2.23 mm/ μ Sec with a standard deviation, S , of 0.037 mm/ μ Sec.

The experimentally derived stress-particle velocity data was fitted to a quadratic form using the principle of least squares. The data points were weighted according to the reciprocal of the product of the errors, i.e., the area of the rectangle occupied by the data on a stress-particle velocity plot. The curve was forced to pass through the origin.

The data was found to be represented by

$$\sigma = 31.290 U_p + 20.786 U_p^2$$

where σ is in kilobars and U_p is in mm/ μ Sec.

Table 1. HUGONIOT DATA FOR MYLAR

<u>Shot</u>	<u>Measured Stress</u> Kbars	<u>Particle Velocity</u> mm/ μ Sec	<u>Shock Velocity</u> mm/ μ Sec	<u>Compression, V/V_0</u>
187	3.55 ± 0.2	0.110 ± 0.001	2.339 ± 0.06	0.953 ± 0.001
188	8.74 ± 0.1	0.238 ± 0.003	2.661 ± 0.03	0.911 ± 0.001
189	14.71 ± 0.5	0.372 ± 0.003	2.865 ± 0.10	0.870 ± 0.004
190	19.95 ± 0.35	0.485 ± 0.008	2.980 ± 0.05	0.837 ± 0.003
191	21.7 ± 0.3	0.519 ± 0.006	3.030 ± 0.04	0.828 ± 0.002
196	27.5 ± 0.6	0.622 ± 0.004	3.204 ± 0.07	0.806 ± 0.004

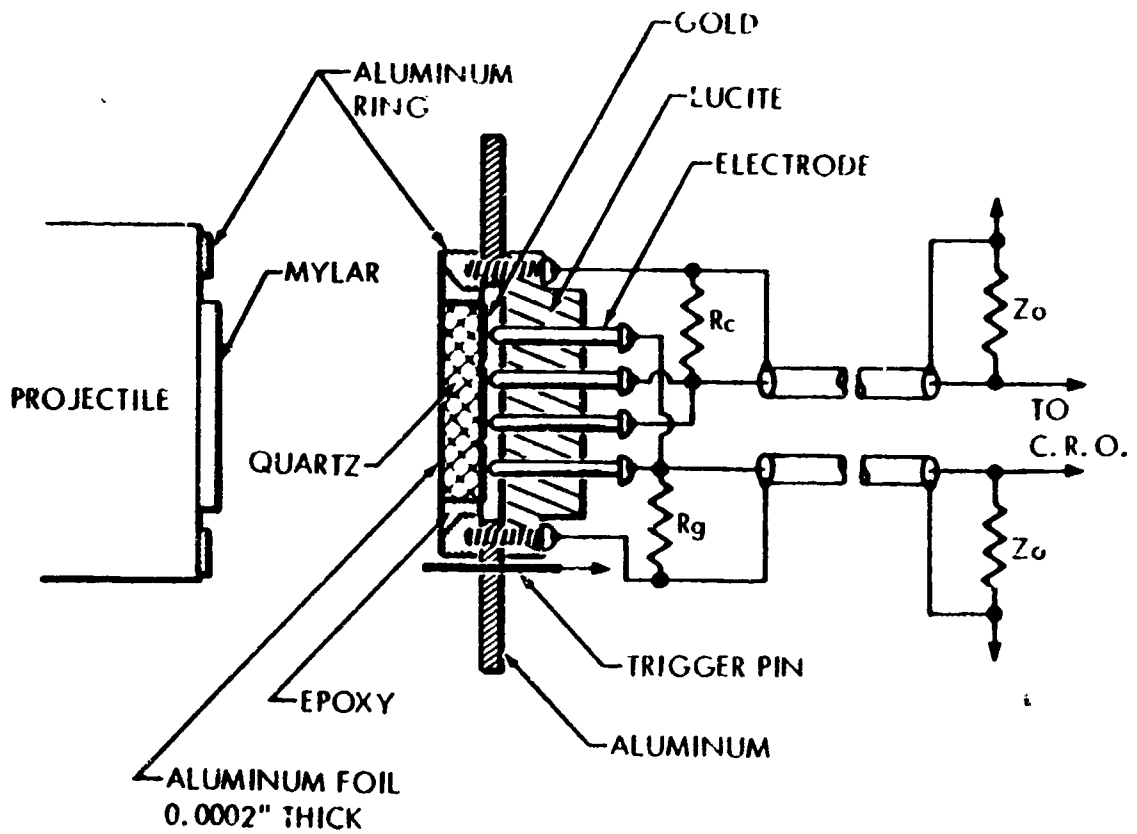


Figure 1 Experimental Arrangement

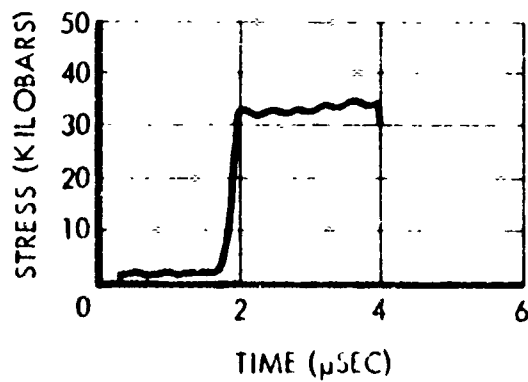


Figure 2. Typical Quartz Gauge Record

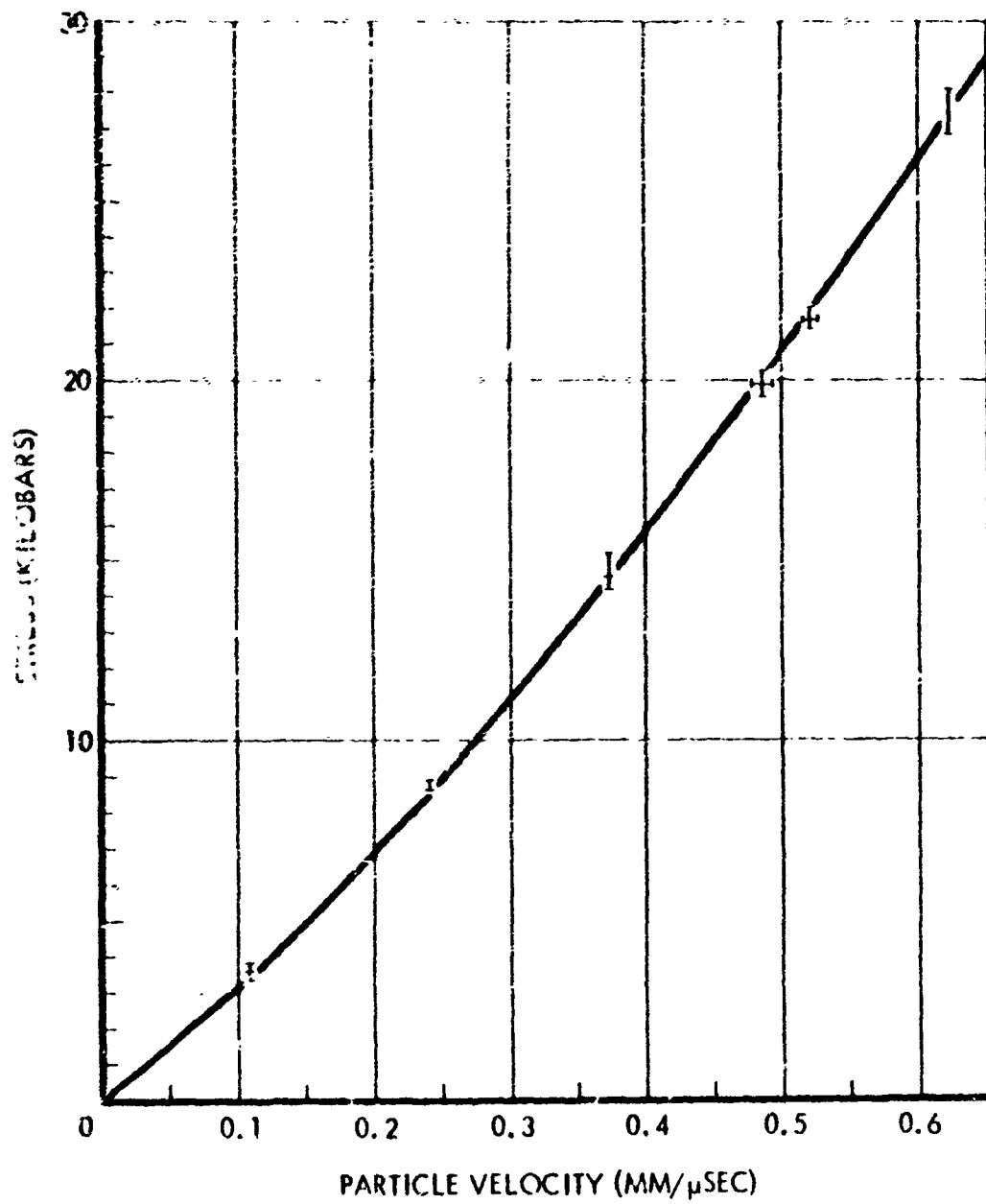


Figure 3. Stress as a Function of Particle Velocity for Mylar

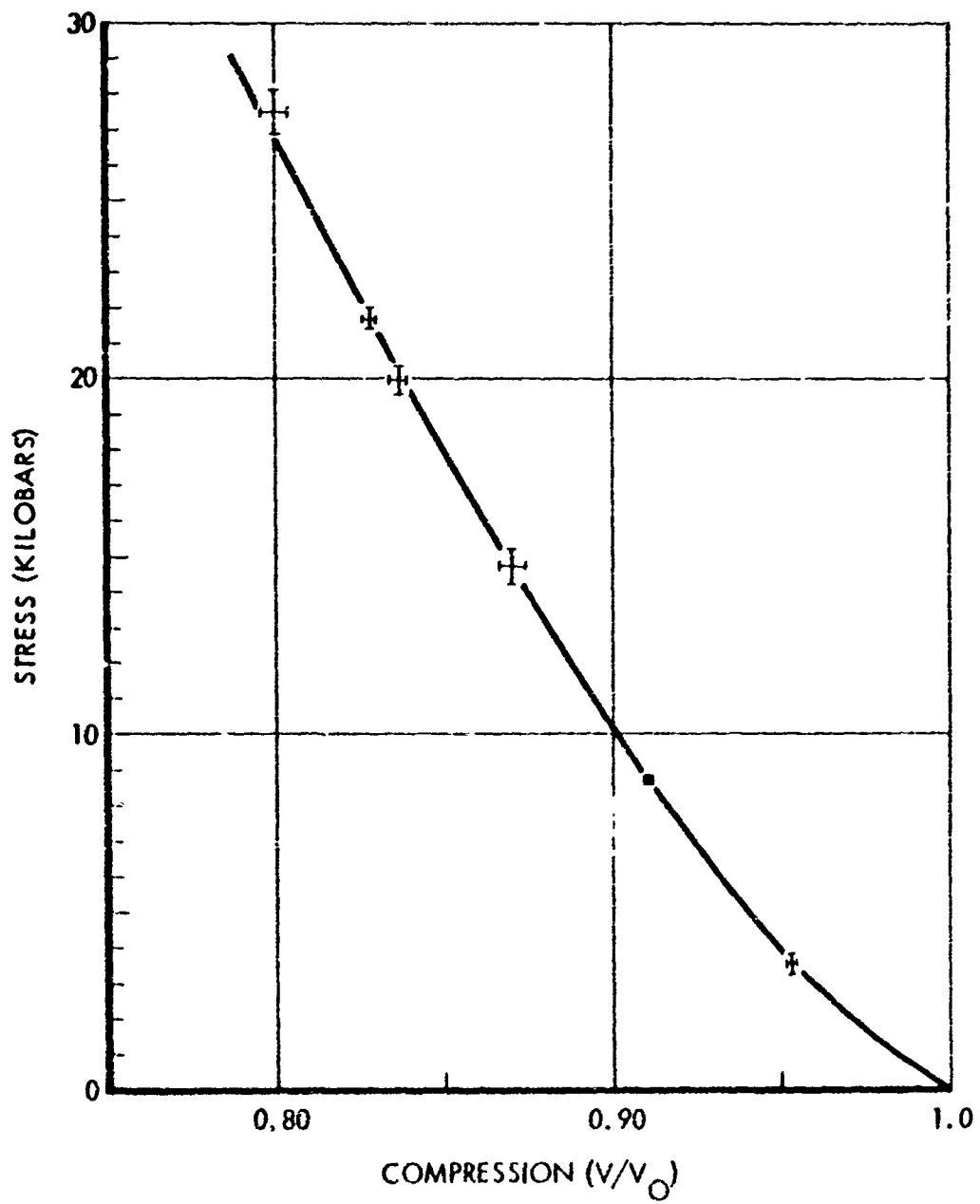


Figure 4. Stress as a Function of Compression for Mylar

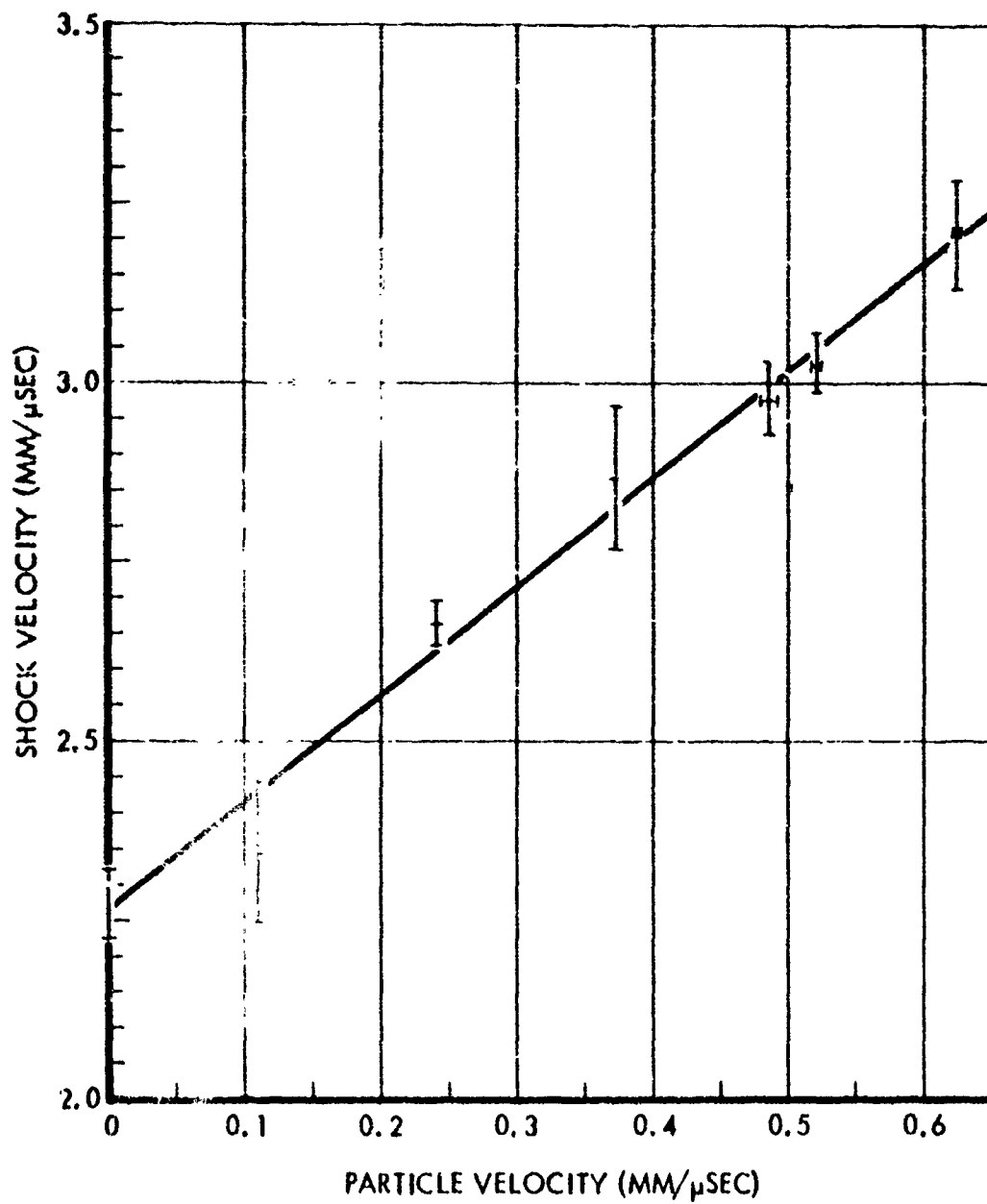


Figure 4. Shock Velocity as a Function of Particle Velocity in Mylar

LIMITATIONS

This document is controlled by _____

All revisions to this document shall be approved by the
above noted organization prior to release.

BOEING

NUMBER
REV LTR

D2-125304-1

ACTIVE SHEET RECORD

SHEET NUMBER	REV LTR	ADDED SHEETS				SHEET NUMBER	REV LTR	ADDED SHEETS			
		SHEET NUMBER	REV LTR	SHEET NUMBER	REV LTR			SHEET NUMBER	REV LTR	SHEET NUMBER	REV LTR
1											
2											
3											
4											
5											
6											
7											
8											
9											
10											
11											

Best Available Copy